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Switching manifold approach to chaos synchronization

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In this Rapid Communication, a switching manifold approach is proposed for synchronizing chaos. The effectiveness of this nonlinear control strategy is demonstrated by both theoretical analysis and numerical simulations on two typical chaotic systems: the Lorenz and the modified Lorenz systems. [S1063-651X(99)51603-6]

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I. INTRODUCTION

The problems of controlling and synchronizing chaos can be formulated under a unified framework [1]. These two subjects have been intensively studied in the last decade [1,2]. Chaos synchronization has many potential applications in laser physics, chemical reactor, secure communication, biomedical science, and so on [3].

It is known that linear or linearized control methods are not always possible for controlling nonlinear systems, and nonlinear control methods prove to be often necessary, especially for chaotic systems [3,4]. Some nonlinear control techniques have even been extended to synchronization of hyperchaos and spatiotemporal chaos [4,5]. One typical method is the variable structure (or sliding mode) control, which has some successful applications for chaotic systems [6].

In this Rapid Communication, we further extend our method [6] to performing synchronization of two chaotic systems with different initial conditions. In addition to theoretical analysis, two Lorenz systems and two modified Lorenz systems are simulated to demonstrate the effectiveness of this method.

Consider two n-dimensional chaotic systems,

$$\dot{x} = F(x), \quad x \in \mathbb{R}^n,$$
 (1)

$$\dot{y} = F(y) + G(x)u, \quad y \in \mathbb{R}^n, \quad u \in \mathbb{R}^m, \quad (2)$$

where *F* is a vector-valued nonlinear function satisfying some defining conditions, and G(x) is an $n \times m$ matrix-valued nonlinear function to be determined along with the controller u=u(x,y).

The goal here is to force the two coupled systems to be synchronized even if they have different initial conditions. As usual, we call system (1) the master system, and system (2), the slave system.

The basic controller design principle is outlined as follows. To start with, a switching manifold containing the desired chaotic target (for synchronization) of the master system, is found. Then, using a nonlinear control strategy, the state of the slave system is driven to move toward the manifold from any nearby place. In much the same way, another switching manifold is obtained for a chaotic state nearby the target, and the trajectory is forced to slide onto it. The control law so designed can be in a very simple nonlinear form. The effectiveness of such a control strategy can be analyzed by both theoretical analysis and numerical simulation, as demonstrated below.

II. ANALYSIS OF SWITCHING MANIFOLDS

To illustrate the proposed control method and design procedure, it is especially convenient to use examples. The wellknown Lorenz system and its modified version [7] are taken as examples for this purpose.

Example 1. Consider two coupled Lorenz systems, where the first system is given by

$$\dot{x}_{1} = -\sigma(x_{1} - x_{2})$$
$$\dot{x}_{2} = \rho x_{1} - x_{2} - x_{1} x_{3}$$
$$\dot{x}_{3} = x_{1} x_{2} - b x_{3},$$
(3)

and the second one has the same form, with x_i being replaced by y_i , i = 1,2,3, respectively, which are assumed to have two sets of different initial conditions.

Example 2. Consider two coupled modified Lorenz systems [7], in which the two product terms x_1x_3 and x_1x_2 in Eq. (3) are replaced by $20x_1x_3$ and $5x_1x_2$, respectively, with two different initial conditions.

To simplify our presentation, we only analyze Example 1 here. By adding a controller into the right-hand side of the first equation of the slave system, we have

$$\dot{y}_1 = -\sigma(y_1 - y_2) + u.$$
 (4)

Then, by subtracting Eq. (3) from the slave system, with the first equation being replaced by Eq. (4), and by defining the synchronization error as

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$$e_i = y_i - x_i, \quad i = 1, 2, 3,$$

we obtain the synchronization error dynamics,

$$\dot{e}_{1} = -\sigma(e_{1} - e_{2}) + u$$

$$\dot{e}_{2} = \rho e_{1} - e_{2} - e_{1}(x_{3} + e_{3}) - e_{3}x_{1}$$

$$\dot{e}_{3} = e_{1}(x_{2} + e_{2}) + x_{1}e_{2} - be_{3},$$

(5)

where, compared with Eq. (2), $G = (1,0,0)^{\top}$. Moreover, let the controller be

$$u = u_{eq} + u_0, \quad |u_0| \le \epsilon, \tag{6}$$

where u_{eq} and u_0 are to be determined, and $\epsilon > 0$ is the allowable bound for the control inputs.

For synchronization purpose, as mentioned above, the first step is to find a suitable switching manifold, and then to design an effective nonlinear control to drive the error state to move toward this stable manifold. In so doing, the error state will eventually approach zero along (or near) the manifold. Select the manifold to be in the form

$$s = s(e), \tag{7}$$

which has to be specified such that (i) it contains the target, e=0, and (ii) $\partial s/\partial u \neq 0$ almost everywhere (near e=0).

For the error dynamical system (5), the desired manifold (usually not unique) can be taken as

$$s(e) = be_3 - e_1^2 = 0, (8)$$

where *b* is a positive constant. For this chosen manifold, it is easy to verify that the two conditions, (i) and (ii) described above, can be satisfied: first, e=0 is contained in the manifold; second, the vector $G=(1,0,0)^{\top}$ is transversal to the manifold at the point e=0. This is because

$$\dot{s} = b[e_1(x_2 + e_2) + x_1e_2 - be_3] - 2e_1[-\sigma(e_1 - e_2) + u],$$
(9)

which implies



FIG. 1. Synchronization between y_1 and x_1 after a transient (2000 steps) for Example 1 with different initial conditions controlled by the controller (13) with $u_{ea} \neq 0$.

$$\frac{\partial \dot{s}}{\partial u} = -2e_1, \tag{10}$$

and this does not vanish along manifold (8) except when $e_1 = 0$. Therefore, when $e_1 \neq 0$, \dot{s} can be directly controlled by u.

Next, we study the dynamical behavior of the error system (5) when it is confined on the manifold by the controller u.

Remark 1. If G is taken as $(0,1,0)^{\top}$, then the controlled Lorenz system becomes

$$\dot{e}_{1} = -\sigma(e_{1} - e_{2})$$

$$\dot{e}_{2} = \rho e_{1} - e_{2} - e_{1}(x_{3} + e_{3}) - e_{3}x_{1} + u \qquad (11)$$

$$\dot{e}_{3} = e_{1}(x_{2} + e_{2}) + x_{1}e_{2} - be_{3}.$$

In this case, a switching manifold can be selected as

$$s(x) = be_3 - e_2^2 = 0. \tag{12}$$

The manifolds (8) and (12) can be used simultaneously for the two examples studied in this paper, as shown below by numerical simulations.



FIG. 2. Total dynamical error $e = \sum_{i=1}^{3} e_i$ for Example 1 with different initial conditions controlled by the same controller (13), as in Fig. 1. In this figure, (a) $u_{eq} = 0$; (b) $u_{eq} \neq 0$. (All of ordinate is dimensionless.)



FIG. 3. Synchronization between y_1 and x_1 after a transient (2000 steps) for Example 2 with different initial conditions controlled by the controller (13) with $u_{eq} \neq 0$. (Ordinate is dimensionless.)

III. CONTROLLER DESIGN PROCEDURE

Recall that the task of control is to force s in Eq. (9) to tend to zero. Therefore, the controller design must follow this principle: to drive all the error states, typically those nearby the manifold, to converge onto the stable manifold.

Consider Eq. (9) again. If all the variables e_1, e_2, e_3 can be directly measured, then the control law can be chosen in the form of Eq. (6), in which u_{eq} is designed to ensure \dot{s} = 0 whenever s = 0. Intuitively, u_{eq} plays the dominant control, whenever u_0 fails to achieve synchronization alone. Observe that $-b^2e_3 = -bs - be_1^2$, so that Eq. (9) can be rewritten as

$$\dot{s} = b[e_1(x_2+e_2)+x_1e_2]-bs-be_1^2+2\sigma e_1(e_1-e_2)-2e_1u$$

Based on this, it is easy to see that we can use

$$u_{ea} = (\sigma - b/2)(e_1 - e_2)$$

and

$$u_0 = \epsilon \operatorname{sgn}[e_1 s];$$

that is,

$$u = (\sigma - b/2)(e_1 - e_2) + \epsilon \operatorname{sgn}[e_1(be_3 - e_1^2)]. \quad (13)$$

To this end, the controlled system (9) with controller (13) becomes

$$\dot{s} = -bs - \epsilon \operatorname{sgn}[e_1s] + d(t), \tag{14}$$

where $d=b(e_1x_2+x_1e_2)$ may be viewed as a disturbance. Because this type of controller is robust, such disturbance can be attenuated, as well documented in the conventional nonlinear control literature.

Remark 2. Even if $\epsilon = 0$ is used in the control law, namely,

$$u = u_{eq} = (\sigma - b/2)(e_1 - e_2), \tag{15}$$



FIG. 4. Total dynamical error $e = \sum_{i=1}^{3} e_i$ for Example 2 with different initial conditions controlled by the same controller (13), as in Fig. 3. In this figure, (a) $u_{eq} = 0$; (b) $u_{eq} \neq 0$. (All of ordinate is dimensionless.)

synchronization is still possible. In this case, however, the robustness enhanced by the additional control term u_0 is generally lost. With a suitable $\epsilon > 0$, the convergence rate can be significantly improved.

Remark 3. If system (11) is considered along with the manifold (12), the following control law works as well:

$$u = u_{eq} + \epsilon \operatorname{sgn}[e_2(be_3 - e_2^2)], \qquad (16)$$

$$u_{eq} = e_1[e_3 - (\rho - b/2)] + (1 - b/2)e_2.$$
(17)

This is similar to the controllers (13) and (15).

Remark 4. If $u_{eq}=0$ is used in the controller, then the above control task fails, at least in our simulations.

Remark 5. If only a simple linear feedback, $u = -ke_i$, k > 0, i = 1 or 3, is used, the above control task also fails in our simulations.

IV. SIMULATION AND DISCUSSION

The Lorenz system has a familiar chaotic attractor for the parameters set $\sigma = 10$, b = 8/3, and $\rho = 28$, whereas the modified Lorenz system has a chaotic attractor for $\sigma = 16$, b = 4, and $\rho = 45.92$.

To demonstrate the effectiveness of the developed control method, we have studied various numerical simulations for synchronization between two identical Lorenz systems (see Examples 1 and 2) with different initial conditions.

Figure 1 shows the dynamical behavior of the synchronization between y_1 and x_1 after a transient (2000 steps) for Example 1 with different initial conditions: $x(0) = (0, -1, 0)^{\top}$ and $y(0) = (0.05, -0.05, 0.01)^{\top}$. The controller is given by Eq. (13), namely,

$$u = u_{eq} + u_0 = (\sigma - b/2)(e_1 - e_2) + \epsilon \operatorname{sgn}[e_1(be_3 - e_1^2)].$$

Figure 2 shows the total dynamical error $e = \sum_{i=1}^{3} e_i$, associated with Fig. 1. In Fig. 2(a), $u_{eq} = 0$; in 2(b), $u_{eq} \neq 0$.

Figure 3 shows the synchronization behavior between y_1 and x_1 after a transient (2000 steps) for Example 2 with different initial conditions: $x(0) = (0.01, -0.01, 0.05)^{\top}$ and $y(0) = (0.05, -0.05, 0.01)^{\top}$. The controller is also given by Eq. (13).

Figure 4 shows the total dynamical error $e = \sum_{i=1}^{3} e_i$, associated with Figure 3. In Fig. 4(a), $u_{eq} = 0$; in 4(b), $u_{eq} = 0$. Simulation results using controller (16) for Examples 1 and 2 are very similar to Fig. 4.

It is clear from Figs. 1–4 that these numerical simulations indeed have verified that the above theoretical analysis is correct and our nonlinear control strategy is effective. More specifically, using either controller (13) or (16), precise synchronization is achieved for the two examples. However, the control fails for synchronizing these two coupled systems, as shown in Figs. 2(b) and 4(b), if $u_{eq} = 0$ is taken in Eq. (13) or (16). The control also fails if only a simple linear feedback $u = ke_i$ (i = 1 or 3) is used. This demonstrates the advantage of the proposed control strategy. The main idea and the method of this Rapid Communication can be extended to other chaotic/hyperchaotic systems in principle.

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